1.	(Currently Amended)	An axiai-now mermai turbomachine naving comprising:
	_a metallic rotor (1), in which	i
	a circumferential groove;	
	_rotor blades (3) -made of an ir	ntermetallic compound are mounted in a the circumferential
groove to form a row of blades , characterized in that ;		
	_at least two rotor blades (3')	which are positioned at a uniform distance from one another
and are made of a more ductile material are more ductile than said intermetallic compound, said		
at least two rotor blades arranged in said row of blades between the intermetallic rotor blades (3),		
1		
	wherein said at least two the	rotor blades (3') made of the more ductile material are either
being considerably		
longer than the intermetallic rotor blades (3) or, if they, or		
	are of the same length	as, having and have a different blade tip shape than the
intermetallic <u>rotor</u> blades (3) .		
2.	(Currently Amended)	The turbomachine as claimed in claim 1, characterized in
that further comprising:		
	_intermediate pieces (4)-made	of a more lightweight material than the material of the rotor
(1), preferably made of an intermetallic compound or a titanium alloy, are additionally arranged		
between two adjacent rotor blades (3, 3') of a row of blades.		
3.	(Currently Amended)	The turbomachine as claimed in claim—1 or 2, characterized

4. (Currently Amended) The turbomachine as claimed in claim 3, characterized in that wherein the γ -titanium aluminide alloy has the following chemical composition (in % by

in that wherein the intermetallic compound for of the rotor blades (3) and the lightweight

material of the intermediate pieces (4) is each comprises an alloy selected from the group

consisting of a γ -titanium aluminide alloy or and an orthorhombic titanium aluminide alloy.

weight): Ti-(30.5-31.5)Al-(8.9-9.5)W-(0.3-0.4)Si.

- 5. (Currently Amended) The turbomachine as claimed in one of claims 1 to 4 Claim

 1, characterized in that wherein the rotor blades comprise blade tips (5) of the rotor blades (3)

 can be coated with a hard phase.
- 6. (Currently Amended) The turbomachine as claimed in claim 5, characterized in that wherein the blade tips each comprise a wear-resistant layer can be applied laser welded to the blade tips by means of laser welding.
- 7. (Currently Amended) The turbomachine as claimed in one of claims 1 to 6Claim 1, characterized in that wherein the turbomachine is comprises a gas turbine having a high-pressure compressor of a gas turbine with a comprising said rotor (1) which substantially comprises, said rotor comprising a stainless Cr-Ni steel.
- 8. (Currently Amended) The turbomachine as claimed in one of claims 1 to 7Claim

 1, characterized in that the wherein said rotor blades (3') which are more ductile than the intermetallic rotor blades (3) consist of comprise a material selected from the group consisting of stainless Cr-Ni steel, or a heat-resistant turbine blade steel-or, and a superalloy.
- 9. (New) The turbomachine as claimed in Claim 2, wherein said lightweight material comprises an intermetallic compound or a titanium alloy.